# ASSESSMENT OF MILITARY AIRFIELD OBSTACLE FREE SPACE BASED ON GIS

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**Abstract:** A mathematical model of obstacle limit surfaces for military airfield obstacle free space is established through airfield obstacle free space analysis. Based on the model, triangle mesh elevation model of military airfield obstacle free space is built by using the software-ArcGIS, and the 3-D display result is obtained. It is convenient to evaluate military airfield obstacle for superimposing digital elevation model (DEM) with military airfield topographic map. Thus it improves the efficiency greatly. It lays the foundation for the application of geographic information systems (GIS) in the management of military airfield obstacle free space.

**Key words:** airfield obstacle free space; mathematical model; 3-D visualization; geographic information system (GIS)

## INTRODUCTION

Airfield obstacle free space is the airspace near the airfield along the route of taking off and landing within a certain range, that is, the airspace need to be on both ends and sides of the runway as the plane takes off to climb, lands to slide and circles for vision. In this region, ground obstacles impeding the navigation and flight cannot exist. Therefore, airfield obstacle free space should be managed and controlled after carrying out the site of airfield or putting into use<sup>[1]</sup>.

There are already some studies on establishing 3-D model of airfield obstacle free space surface. Some models are built by 3DSMAX software. This modeling approach does not contain geographic information data. There are some limitations in application. Some are built by digital elevation model (DEM) in the grid format, but the grid of the data is calculated through interpolation. It is mainly applied to vast territory and small undulating topography surface simulation, and it is not appropriate for the airfield obstacle

free space surface which marks the boundary line and changs largely in slope geometry. So there is also rare related research data at home and abroad at present.

The traditional method of assessing the airfield obstacle free space is to map out the free space limit surfaces and obstacle location, and then calculate whether the height of obstacles exceeds the limit height under the free space requirements and triangular relationship. Mapping in this way is slow, low accuracy, difficult to update. Through the establishment of airfield obstacle free space model, it can determine obstacle permit height in all planar location around the airfield. Processing and analyzing the obstacles data by using geographic information system (GIS) technology can obtain high efficiency, accurate results, and achieve 3-D visualization of airfield obstacle free space.

In order to perform military airfield obstacle free space surface more accurately, manage military airfield obstacles more effectively, and check the location of any surface limit height neatly and quickly, the mathematical model of obstacle limit surfaces is bulit for military airfield obstacle free space and the 3-D model of airfield obstacle free space surface is given.

# 1 CONSTRUCTION OF MATHE-MATICAL MODEL

# 1.1 Airfield obstacle free space specifications and coordinate establishment

Airfield obstacle free space zone is composed with runway strip, end free space zone and side free space zone. Its composition is shown in Fig. 1.

free space limit surfaces are symmetrical vertically and horizontally on the basis of an analysis of airfield obstacle free space specifications, as shown in Fig. 1. Therefore, it can be described by using a quarter model of airfield obstacle free space specifications, as shown in Fig. 2.

It seems that each level of airfield obstacle

In Fig. 2, L is the length of runway strip, a, b, respectively, the length of one end and the width of one side of free space zone. To facilitate processing data, it still considers the midpoint of the runway axis as the coordinate origin, and establishes coordinate systems. Units are unified as  $m^{[1,2]}$ .

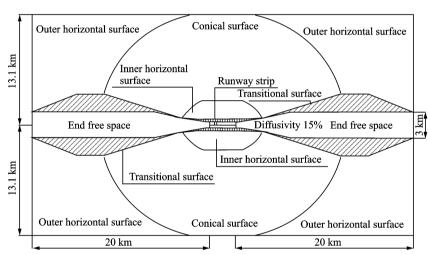


Fig. 1 Composition of airfield obstacle free space surface

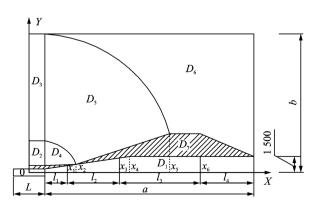


Fig. 2 Quarter model of airfield obstacle free space surface

## 1. 2 Variable setting

In Fig. 2,  $x_1 = l_1 + L/2$ ;  $x_2$  is the X-coordinate value where the inner horizontal surface sideline connects with end free space surface;  $x_3 = l_1 + l_2 + L/2$ ;  $x_4$  is the X-coordinate value when the end free space surface extends to 3 000 m wide;  $x_4 = 9$  333 + L/2;  $x_5$  is the X-coordinate value where the outer horizontal surface connects with end free space transitional surface;  $x_6 = l_1 + l_2 + l_3 + L/2$ . Tables 1, 2 list parameters of limit surface of airfield obstacle free space.

Table 1 Parameters of limit surface of end airfield obstacle free space

The first length			The second length			The third length			The fourth length			
Length	Gradient	End height	Length	Gradient	End height	Length	Gradient	End height	Length	Gradient	End height	Total length
$l_1$	$k_1$	$H_1$	$l_2$	$k_2$	$H_2$	$l_3$	0	$H_3$	$l_4$	$k_4$	$H_w$	а

Table 2 Parameters of limit surface of side airfield obstacle free space

Transitional surface	Inner horizo	ontal surface		Conical	_ Total width of each side		
Gradient	Radius	Height	Radius	Gradient	Outer sideline height	of runway midline	
1/10	$r_1$	$H_n$	$r_2$	k'	$H_w$	b	

## 1.3 Transitional surface measurements

For measurements of the inner and outer horizontal surfaces, conical surface, and end free space surface, airfield obstacle free space specifications are defined, but the measurement of the transitional surface need to be calculated. The

specifications provide that transitional surface be-

gins with end free space and sidelines of runway

strip, tilts outward with gradient 1/10, until it

intersects the side free space surface. In Fig. 3, for point A in X-axis, define vari-

able x as its abscissa value. Over point A make

the X-axis vertical line and intersect the sidelines of end free space at B. Set the length of AB as  $d_1$ , and the height of A, B point free space as  $h_1$ . The starting point of the transitional surface corresponding to point B is E. From point E make a line perpendicular to X-axis with gradient 1/10 until intersects side free space conical surface at F, and set the height of the point free space as  $h_2$ . The projection of point F in horizontal surface is C. Set the distance between point C and the runway axis as  $d = d_1 + d_2$ . The length of BC is the width of the transitional surface, set it as  $d_2$ .

Through analysis of free space measurements,  $h_1$  is the sub-function of x

Define a variable r as the distance between point

C and the origin,  $r^2 = (x - L/2)^2 + y^2$ .

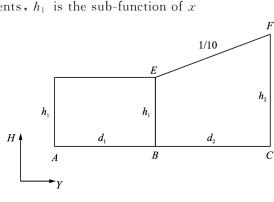


Fig. 3 Calculation principle of width of transitional surface

$$h_{1}=f_{1}(x)= \\ \begin{cases} 0 & 0 < x \leq L/2 \\ k_{1}(x-L/2) & L/2 < x \leq x_{1} \\ H_{1}+k_{2}(x-x_{1}) & x_{1} < x \leq x_{3} \\ H_{2}=H_{3} & x_{3} < x \leq x_{6} \\ H_{3}+k_{4}(x-x_{6}) & x_{6} < x \leq a+L/2 \end{cases}$$
 (1)

 $h_2$  is the permit height of circular side free space, and it is the sub-functions of the radius r  $h_2 = f_2(r) =$ 

$$f_{12} = f_{2}(r) =$$

$$\begin{cases}
H_{n} & r \leqslant r_{1}, & L/2 < x \leqslant x_{2} \\
k'(r - r_{1}) + H_{n} & r_{1} < r \leqslant r_{2}, & x_{2} < x \leqslant x_{5} \\
H_{w} & r_{2} < r, & x_{5} < x \leqslant a + L/a
\end{cases}$$

The relationship between  $d_1$  and x is  $d_1 = f_3(x) =$ 

 $(100+0.15(x-L/2) L/2 < x \le x_4$ 

1 500 
$$x_4 < x \le a + L/2$$
  
The width of the transitional surface is

 $d_2 = 10(h_2 - h_1) \tag{4}$ 

$$d_2 = 10(h_2 - h_1) \tag{4}$$

Then

 $d_2 =$ 

$$r^{2} = (x - L/2)^{2} + (d_{2} + d_{1})^{2}$$

$$d = d_{1} + d_{2}$$
(6)

$$d = d_1 + d_2 \tag{6}$$

In accordance with sub-points, substituting Eqs. (3,5,6) into Eq. (4)

$$100 + 10H_{n} \qquad 0 < x \le L/2$$

$$10 \left[ H_{n} - k_{1} (x - L/2) \right] \qquad L/2 < x \le x_{1}$$

$$10 \left[ H_{n} - H_{1} - k_{2} (x - x_{1}) \right] \qquad x_{1} < x \le x_{2}$$

$$10 \left[ k' \left( \sqrt{(x - L/2)^{2} + \left[ d_{2} + 100 + 0.15 (x - L/2) \right]^{2}} - r_{1} \right) + H_{n} - H_{1} - k_{2} (x - x_{1}) \right] \qquad x_{2} < x \le x_{3}$$

$$10 \left[ k' \left( \sqrt{(x - L/2)^{2} + \left[ d_{2} + 100 + 0.15 (x - L/2) \right]^{2}} - r_{1} \right) + H_{n} - H_{2} \right] \qquad x_{3} < x \le x_{4}$$

$$10 \left[ k' \left( \sqrt{(x - L/2)^{2} + (d_{2} + 1500)^{2}} - r_{1} \right) + H_{n} - H_{3} \right] \qquad x_{4} < x \le x_{5}$$

$$10 (H_{m} - H_{1}) \qquad x_{5} < x \le x_{6}$$

 $x_6 < x \leqslant a + L/2$ 

 $10[H_{20} - H_3 - k_4(x - x_6)]$ 

(7)

(3)

lated.

surface[3].

 $y \in (0,150 \ 0]$ ;

 $(d, r_2 \rceil);$ 

(d,b];

H(x,y) =

(y-100)/10

 $(y-1500)/10+H_2$ 

According to the model it can be obtained free space limit measurements of any part of the

# 1.4 Establishing mathematical model

of obstacles H(x, y) is a sub-function, it can be divided into seven parts  $D_1$ ,  $D_2$ ,  $D_3$ , ...,  $D_7$ 

From Fig. 2 we can see that the limit height

$$D_1$$
:  $\{x, y \mid x \in (L/2, x_4], y \in (0, 100 + 0.15(x-L/2)]\} \cup \{x, y \mid x \in (x_4, a + L/2],$ 

$$H(x,y) = h_1 = f_1(x)$$
 (8)

$$D_2: \{x, y \mid x \in (0, L/2], y \in (100 + 10H_n, r_1]\};$$

$$H(x, y) = H_x$$
(9)

$$D_3: \{x, y | x \in (0, L/2], y \in (r_1, r_2] \};$$

$$H(x, y) = H_n + k'(y - r_1)$$
(10)

$$H(x,y) = H_n + k'(y - r_1)$$
(10)  
$$D_4: \{r, x, y | r \in (0, r_1], x \in (L/2, x_2], y \in (L/2, r_1) \}$$

$$H(x,y) = H_n \tag{11}$$

$$D_5: \{r, x, y | r \in (r_1, r_2], x \in (L/2, x_5], y \in (d, r_2]\};$$

$$\{d, r_2 \rfloor \}$$
;  $H(x,y) = H_n + k' [\sqrt{(x-L/2)^2 + y^2} - r_1]$ 

$$D_6: \{r, x, y | r \in (r_2, a], x \in (0, a + L/2], y \in [r_2, a] \}$$

$$H(x,y) = H_w$$
 (13)  
$$D_7: \{x, y \mid x \in (0, L/2 \rceil, y \in (100, 100 + L/2)\}$$

$$= (100, 100 +$$

$$10H_n$$
]  $\cup \{x,y | x \in (L/2,a+L/2], y \in (d_1, d_2]\};$ 

$$\{a_1,y\in(d_1,d_2]\};$$

$$[y-100-0.15(x-L/2)]/10+k_1(x-L/2)$$

$$L/2 < x \le x_1$$
 $[y-100-0.15(x-L/2)]/10+H_1+k_2(x-x_1)$ 
 $x_1 < x \le x_2$ 

$$[y-100-0.15(x-L/2)]/10+H_2$$
 $x_3 < x \le x_4$ 

 $0 < x \leq L/2$ 

 $x_4 < x \leq x_6$ 

$$(y-1500)/10+H_2+k_4(x-x_6)$$

$$x_6 < x \le a+L/2$$
(14)

then the limit height of the obstacle can be calcu-

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# 2. 1 Implementation process

cy, triangle(TIN) format DEM modeling is used, and the software platform is AutoCAD2006 and ArcGIS family software developed by ESRI<sup>[3]</sup>.

In order to achieve a certain degree of accura-

A brief description of the implementation process as follow:

(1) Draw out the airfield free space surface layout using AutoCAD, and then convert the line file by using ArcCatalog into Polyline type of

shapefile format, as shown in Fig. 4.

shown in Fig. 5.

(2) Edit generated shapefile files using ArcInfo. According to the above mentioned mathematical model of airfield free space, pick points

face and enter the elevation in the property table manually. Define the edges as force line, and generate DEM of TIN format using 3-D analysis, as

from the defined sidelines in every free space sur-

(3) Open ArcScene, and load generated TIN file, then 3-D modle can be obtained, as shown in Fig. 6<sup>[4]</sup>.

(4) Overlay military airfield obstacle limit surfaces to relevant airfield topographic map, as shown in Fig. 7.

Airfield free space DEM overlay model established by using the above method contains corresponding elevation data. It can find the obstacle

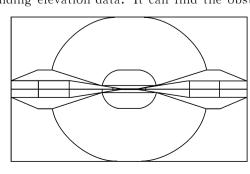


Fig. 4 Plan of military airfield free space surface

According to Eq. (14), the zone the obstacle belongs to is estimated through coordinates, and

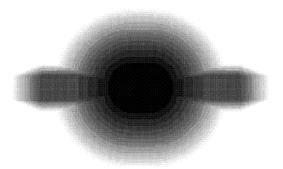


Fig. 5 DEM of TIN format

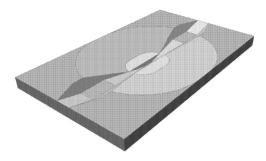


Fig. 6 3-D display

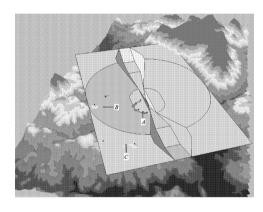


Fig. 7 Military airfield obstacle limit surfaces and relevant airfield topographic map

limit height anywhere through software query function. If the topographic map of corresponding military airfield is overlayed, it can be very convenient to assess the military airfield obstacle free space zone, thereby improve efficiency greatly [53].

### 2. 2 Case analysis

As shown in Fig. 7, assess airfield obstacle free space for a military airfield, A,B,C are obstacles in inner horizontal surface, outer horizontal surface and conical surface within the airfield obstacle free space respectively. According to elevation values, they are displayed by stretching. These three points are beyond the free space limit surface. The information of these points can be gained using the inquiry function. Query results are shown in Fig. 8. The left boxes show related layer information of the query point, and the right show detail information. The actual elevations of A,B,C are 72,302,407 m. The heights of free space limit surfaces are 60,286,380 m, exceeding 12,16,27 m, respectively.

### 3 CONCLUSION

Military airfield clearance specification is analyzed, and the mathematical model of obstacle limit surfaces for military airfield clearance is built. Based on the model, the triangle mesh elevation model of military airfield clearance is built by using ArcGIS, and the 3-D display result is obtained. Military airfield clearance obstacle is evaluated by superimposing DEM with corresponding topographic maps. Analysis result of a

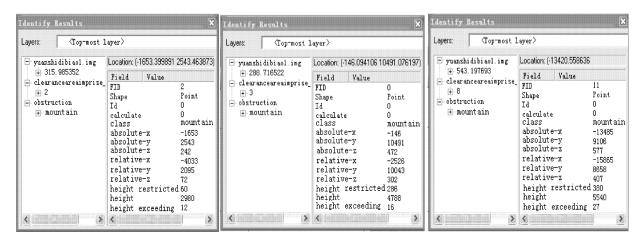


Fig. 8 Obstacle query results

military airfield indicates that the actual elevations of obstacles A, B, C are respectively 72, 302, 407 m, the heights of clearance limit surfaces are 60, 286, 380 m, and the exceeding heights are 12, 16, 27 m, respectively. So, the obstacles of military airfield clearance are evaluated quickly and exactly by the software ArcGIS, and the efficiency of evaluating clearance obstacles can improve greatly.

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## 基于GIS 的机场净空障碍物评定研究

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摘要:通过对军用机场净空规格进行分析,建立了军用机场 净空区障碍物限制面的数学模型。在此数学模型的基础上 使用 ArcGIS 系列软件建立军用机场净空限制面三角网高 程模型,并进行三维显示。将DEM 与某军用机场地形图相 叠加,则可以很方便的对军用机场净空障碍物进行评定,从 而提高工作效率,为使用地理信息系统进行军用机场净空 障碍物管理提供帮助。

关键词:机场净空;数学模型;三维可视化;地理信息系统 中图分类号:V351.11

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