

# Flight Schedule Recovery under Uncertain Airport Capacity

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**Abstract:** Airlines adjust their flight schedules to satisfy more stringent airport capacity constraints caused by inclement weather or other unexpected disruptions. The problem will be more important and complicated if uncertain disruptions occur in hub airports. A two-stage stochastic programming model was established to deal with the real-time flight schedule recovery and passenger re-accommodation problem. The first-stage model represents the flight re-timing and re-fleeting decision in current time period when capacity information is deterministic, while the second-stage recourse model evaluates the passenger delay given the first-stage solutions when one future scenario is realized. Aiming at the large size of the problem and requirement for quick response, an algorithmic framework combining the sample average approximation and heuristic method was proposed. The computational results indicated that the proposed method could obtain solutions with around 5% optimal gaps, and the computing time was linearly positive to the sample size.

**Key words:** flight recovery; passenger re-accommodation; two-stage stochastic model; sample average approximation; heuristic method

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## 0 Introduction

In airline daily operations, the flight irregularity is intrinsically inevitable due to dynamic environment. The uncertainty of weather and other events reduce the airport capacity, which leads to airport congestion and incurs flight delay and cancellation. Recovery policies could be implemented to mitigate the disruption. In 2013, the average on-time ratio was 78.4% in the U. S. according to the 16 main carriers' data from Bureau of transportation statistics (BTS). Each irregular flight costs \$16,600 on average, including expenses on fuel, maintenance, crew, passenger time and welfare loss. The situation is nothing but worse in China. According to the official statistic, the number of irregular flights was up to around 769 000, and the average punctuality ratio was only 72.34% in 2013, which meant about 2 100 irregular flights waited for being dealt with

every day. With the speedy development of air transportation in China, the disequilibrium between increasing flight demand and the relative stable air transportation capacity is prominent. The airport slot, which is the time allocated for an aircraft to land or take off, has become a scarce resource in recent years<sup>[1]</sup>.

There are two flight network of airlines: City pair and hub-and-spoke. In city pair network, the disruptions on airports are less troublesome because most passengers travel in itineraries without transferring flights. However, in hub-and-spoke flight network, since large-scale of flights are operated and passengers transfer flights in the hub, disruptions may spread to following flights. The slot resources in hub airports, such as Beijing, Shanghai, have already run out for regular operations. When inclement weather or other unexpected events reduce the airport capacity, the aviation authority (such as FAA, CAAC etc.)

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provides a set of slots for each airline at the disrupted station, and a free flight program allows an airline to assign legs to the set of slots<sup>[2]</sup>.

There are four sub-problems for flight recovery, dealing with the flight timetable, aircraft, crews and passengers. The schedule recovery problem (SRP) is to re-time the flight schedule and determine the cancellation, so it is the basic work for the other three sub-problems. The aircraft recovery problem (ARP) re-assigns aircraft to re-scheduled routings and keeps fleet balance for flights beyond recovery period. The crew recovery problem (CRP) generates new duties for crews. The passenger recovery problem (PRP) re-accommodates disrupted passengers to new itineraries to deliver them to their destinations.

Since aircraft are the most important resources for airlines, ARP attracts most academic focus. Jarrah and Yu researched the network flow model with the aircraft shortage<sup>[3]</sup>. Yan and Yang studied the flight schedule recovery problem with airport temporary closure<sup>[4]</sup>. Argüello, Bard and Yu discussed the flight schedule recovery problem with temporary shortage of aircraft, and applied GRASP algorithmic framework to rearrange aircraft routings<sup>[5]</sup>. In reference to PRP, Bratu and Barnhart studied the flight delay, cancellation decision and the passenger re-assignment, considering the passenger arrival delay cost<sup>[6]</sup>. Zhang and Hansan researched schedule and passenger recovery for a one-stage hub-and-spoke network, considering both arrival and departure capacity constraints<sup>[7]</sup>. Bisailon et al. studied passenger reassignment problem combining fleet assignment and aircraft routings, they designed a large neighborhood search heuristic method to identify improved solution while retaining feasibility<sup>[8]</sup>. The above references concentrated on individual recovery, regarding aircraft, crews, and passengers separately. Recently, more research emphasizes on integrated recovery. Abdelghany studied integrated decision-making tools for flight recovery problem, and put forward the integrated recovery solution with all flight resources involved<sup>[9]</sup>. Petersen et al. are

known as the first team who the fully integrated recovery formulation and approach with computational results presented<sup>[10]</sup>.

The above assume that disruptions are known before decision-making. However, the significant inherent uncertainties make the recovery problems more complicated. Recovery plans from deterministic model result in lack of robustness and more operational costs. In recent years, stochastic programming shows its advantages in many industries such as transportation, manufacturing, finance, and logistics etc. Although few works have been published on airline stochastic recovery problem, there are some research on other aviation fields. Considering the stochastic scheduling of airlines, Rosenberger et al. worked on the simulation software that controlled the uncertain delay time in airline operations<sup>[11]</sup>. Wu explored the inherent delays of airline schedules resulting from limited buffer times and stochastic disruption in airline operations, and the results indicated that airline schedules must consider the stochasticity in daily operations<sup>[12]</sup>. Yen and Birge established a two-stage stochastic integer model on airline crew scheduling problem under uncertain disruptions. They designed a flight-pair branching algorithm<sup>[13]</sup>. Silverwood<sup>[14]</sup> and Karacaoglu<sup>[15]</sup> reviewed the application of stochastic programming techniques on airline scheduling. They indicated that stochastic programming techniques were able to improve the delay recovery performance of the schedule. Sölveling studied the stochastic programming methods for scheduling of airport runway operations. He established a two-stage stochastic integer model, and used sample average approximation (SAA) method and Lagrangian decomposition to solve the model. He also proposed an improved stochastic branch and bound algorithm<sup>[16]</sup>. Referring to the stochastic methods in airlines operational field, Mou and Zhao built an uncertain programming model with chance constraint, and solved it based on classic Hungarian algorithm to deal with the recovery problem under stochastic flight time<sup>[17]</sup>. Arias et al. proposed a combined methodology u-

sing simulation and optimization techniques to cope with the stochastic aircraft recovery problem<sup>[18]</sup>. Guimarans et al. solved the stochastic aircraft recovery problem using large neighborhood search metaheuristic approach combined with simulation at different stages. The results shows stochastic approach performs better than a deterministic approach<sup>[19]</sup>.

Among those stochastic recovery references, uncertainties were reflected only in the flight times. However, it is stochastic external factors where irregular flights come from in most cases. We studied the flight recovery problem combining re-timing, re-fleeting and passenger re-accommodation when the hub undergoes stochastic decreasing hub capacity.

## 1 Problem Statement

The essence target of recovery policies is to re-accommodate passengers as soon as possible when irregularity happens. Every recovery plan should maintain flow balance for every plane, crew, and passenger flow. The basic strategies to recover flights are delay and cancellation. Every flight has its scheduled time of departure(STD) and scheduled time of arrival(STA), denoting the original scheduled times of departure and arrival, respectively. After delay and cancellation applied, every active (not cancelled) flight will have its estimated time of departure(ETD) and estimated time of arrival(ETA), which are estimated times of departure and arrival, respectively. To some minor disruptions, delay might be the intuitive recovery policy, and it may be effective if the delay time is acceptable and will not break the flow balance in the system. However, in many cases, "only delay" may not be a good strategy because the delay may propagate in the following flights, and will cause passenger or crew misconnection. Cancellation is a quick response to recover the flight schedule, but it is costly because a bunch of passengers will be re-accommodated or spilled, and it may also break the aircraft or crew connections. Cancelling a flight leg usually requires rerouting the aircraft, crew and passenger

flows. Since crews and passengers can fly on other legs or even use other transportation mode, there are many ways to maintain their flow balance. Unfortunately, rerouting the plane is more difficult, and the airline may cancel additional legs on the plane's route<sup>[2]</sup>. So when cancellation policy is applied, the controllers usually cancel a flight cycle, which is a sequence of legs that begins and ends at the same airport, to maintain aircraft flow balance at airports.

For aircraft recovery, there are some specific strategies. Aircraft swap or type substitution may be applied to find the possible swap opportunities in the same aircraft type or between other types. Reserved aircraft can be used to solve the disruptions caused by shortage of aircraft, but they are not always stand by for economic concern. Ferry is also a backup but least used strategy, which flies to specific station without passengers in order to perform the following flights.

As mentioned before, when inclement weather or other unexpected events emerge, airlines will get the airport capacity information from air traffic control (ATC) authority in terms of available slots in a unit time period (or time stage if there is no ambiguity). Fig. 1 illustrates the capacity information and three time periods involved in the problem. Usually the slot information can just be known for the current time period, such as time period 1 in the figure. For the later time periods, such as time period 2—4, the discrete probability distribution can be introduced to the capacity scenarios.

Disruption time period  $T_0$  consists of time stages when capacity is reduced.

Recovery time period  $T_1$  is the time period when flight schedule and fleet assignment are rebuilt. Once the fleet is determined, it is trivial to assign specific airplane. The recovery process begins at the very beginning of  $T_1$ , and all flights that scheduled to departure in  $T_1$  are adjustable. Beyond recovery time period, the flight schedule should go back to normal status. It means, all flights that scheduled to departure beyond  $T_1$  should not be delayed nor cancelled, and will be operated by the original aircraft type. It is obvi-

ously that the longer  $T_1$  is, the more adjustable flights are involved, the more possible recovery chances are generated, and the higher the computational complexity will be. The shorter recovery time period forces more usage of cancellation, which guarantees the flight can be recovered to normal quite soon, while it may incur more cost.

Re-accommodation time period  $T_2$  is the time period when passengers can be re-accommodated. It can cover the whole day along, and even can be extended to a longer time if serious disruption happens. Theoretically, all the disrupted passengers can be re-accommodated in the future if the  $T_2$  is long enough, since flights are frequently operated in flight schedule.

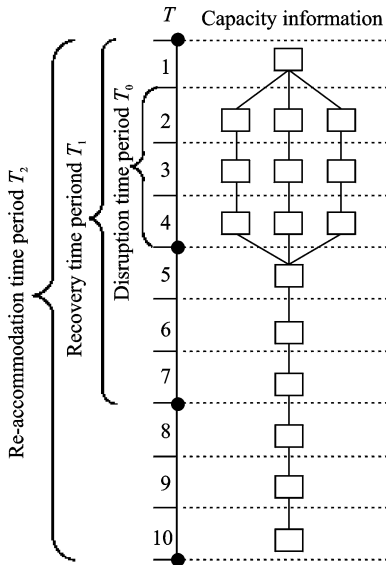


Fig. 1 Illustration on time periods involved and capacity information

During  $T_0$ , the airport capacity is uncertain. In Fig. 1, for example, there are three capacity scenarios, each of which has corresponding probability. Solutions should be found for the current time period before one of the scenarios is realized in the future.

Since flight cycles are considered as units in the recovery process, different types of flight cycles are defined in order to distinguish the different roles in recovery process. Fig. 2 shows the four types of flight cycles, although all flight cycles simply consist of only two flights in the fig-

ure, they can consist more flights in our model. In Fig. 2, A, B, C, D, H represent airports.  $H \rightarrow B$  means the flight F11 departs from airport H and arrives at airport B. Notice that the flight cycle confines to the simple cycle without sub-cycle. Cycles of type (1), (2) and (3) are those with STD in the recovery time period, so they are adjustable in the recovery process. The ETA of type (1) is in the recovery time period, so it can be delayed, cancelled or re-fleeted. The last flight such as F22 in type (2) has STD in the recovery time period, so it is suitable to all the recovery policies as well. Notice that although F22 has STA after the recovery period, it might be delayed to some degree, which will be constrained by delay limitation or aircraft balance requirement in the model. Type (3) has flights with STD after recovery period, so in order to guarantee the flights unchanged, the cycles cannot be cancelled, and the fleet type cannot be changed. Since type (3) has flight like F31 which has STD before recovery period, it can be delayed to some degree as long as it does not influence the regular operation of F32. Cycles of type (4) are those with STD after recovery time period, so they are supposed to fly regularly, and cannot be cancelled, delayed or re-fleeted.

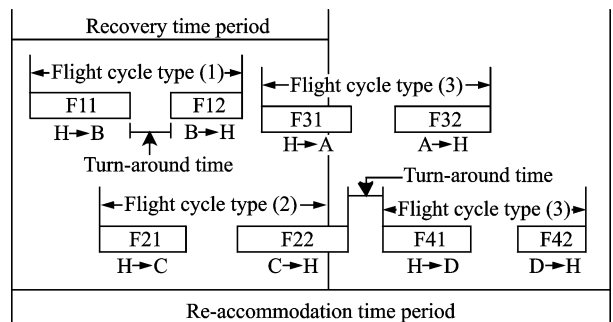


Fig. 2 Four types of flight cycles

## 2 Stochastic Model for Real-Time Integrated Recovery Problem

A two-stage stochastic model was established to deal with the flight schedule problem under uncertain reduced capacity in the hub. The classic

two-stage stochastic model was proposed by Dantzig<sup>[20]</sup> and Beale<sup>[21]</sup>. The first-stage model was designed to find solutions before the uncertain parameters were identified, and the second-stage model presented possible recourse solutions after all the uncertainties were identified. In modeling the recovery problem, assume that the flight network has  $N_{\text{apt}}$  airports, one of which is the hub airport, where all the involved flight cycles begin and end. The slots resources in the spoke airports are sufficient. It is reasonable because in hub-and-spoke network, the aircraft movements in spoke airports are much lesser than those in the hubs.

## 2.1 The first-stage model

The first-stage model determines the recovery decisions in the current time stage. Since the flight timetable and aircraft routing are considered simultaneously, there are 4 candidate recovery policies: delay, cancel, fly and fleet re-assignment.

(1) Set and parameters

$FC = \{1, \dots, f_{FC}\}$  : Set of flight cycles, indexed by  $i$ .

$FC_1$  : Set of adjustable flight cycles, with STD in current time stage.  $FC_1 \subset FC$ .

$F$  : Set of flight legs, indexed by  $f$ .

$\epsilon$  : Set of fleet types, indexed by  $e$ .

$\Omega$  : Discrete disruption scenario set, indexed by  $\omega$ .

$a_e$  : Number of available aircraft of type  $e$  at the hub airport currently.

$c_i^{\text{cancel}}$  : Cancellation cost for flight cycle  $i$ .

$c_i^e$  : Cost of assigning fleet type  $e$  to flight cycle  $i$ .

(2) Decision variables

For each flight cycle  $i \in FC_1$ ,

$w_i$  : Cancellation variable, equals 1 if  $i$  is cancelled, otherwise 0.

$u_i$  : Delay variable, equals 1 if  $i$  is delayed to the next time stage, otherwise 0.

$v_i$  : Fly variable, equals 1 if  $i$  flows in current time stage, otherwise 0.

$k_i^e$  : Fleet assignment variable, equals 1 if type  $e$  is assigned to  $i$ , otherwise 0.

Let  $x$  denote the above decision variables in the first-stage model for expressional simplicity

$$\min Z = \sum_{i \in FC_1} (w_i \cdot c_i^{\text{cancel}} + \sum_{e \in \epsilon} k_i^e \cdot c_i^e) + E[Q(x, \omega)] \quad (1)$$

$$w_i + u_i + v_i = 1, \forall i \in FC_1 \quad (2)$$

$$\sum_{i \in FC_1} v_i \leq D_1 \quad (3)$$

$$\sum_{e \in \epsilon} k_i^e = v_i, \forall i \in FC_1 \quad (4)$$

$$\sum_{i \in FC_1} k_i^e \leq a_e, \forall e \in q \quad (5)$$

In Eq. (1),  $E[Q(x, \omega)]$  represents the expectation for given  $x$  under uncertain capacity parameters. The objective function of Eq. (1) is to minimize the cancellation, re-fleeting costs plus the expected passenger arrival delay. The first-stage constraints are defined to choose the schedule recovery and re-fleeting decisions on the flight cycles of current time stage. Constraints in Eq. (2) require that one decision must be made for the flight cycles that in the first-stage, they should fly, be cancelled or delayed. Constraint in Eq. (3) represents the departure capacity restriction in current time stage on the hub airport.  $D_1$  is a deterministic capacity parameter in the model. The arrival capacity limitation is not considered in this model because fly time for a flight is assumed to be constant, so the arrival time of the relative flights should be already known at current time stage. Constraints in Eq. (4) are the fleet assignment constraints. For those flown flight cycles, one fleet must be assigned. Constraints in Eq. (5) are aircraft resource constraints, which indicate that aircraft of each fleet type are limited. The total number of aircraft that assigned to those flown flight cycles in current time stage cannot exceed the available amount of aircraft for each fleet type.

## 2.2 The second-stage resource model

The second-stage resource model reflects the expected passenger delay considering the future flight schedule recovery. In Eq. (1),  $E[Q(x, \omega)]$  represents the expectation for given  $x$  under uncertain capacity parameters. The uncertainty can be modeled as discrete disruption scenario set  $\Omega$ ,

and each  $\omega \in \Omega$  has its corresponding probability  $p_\omega^r$ . Thus  $E[Q(x, \omega)]$  can be computed by optimization problem  $Q(x, \omega)$ .

(1) Set and parameters

$P$ : Set of passenger itineraries, indexed by  $p, r$ .

$\lambda_f^i$ : Equals 1 if flight leg  $f$  is in flight cycle  $i$ , otherwise 0.

$SEAT_e$ : Seat number of fleet type  $e$ .

$\delta_f^r$ : Equals 1 if flight leg  $f$  is in itinerary  $r$ , otherwise 0.

$n_p$ : Number of passengers on itinerary  $p$ .

$d_p^r$ : Total cost of delay incurred when itinerary  $p$  passengers is re-accommodated on itinerary  $r$ .

$R(p, x, \omega)$ : Set of candidate recovery itineraries for each itinerary  $p$  based on first-stage decision  $x$  under scenario  $\omega$ , which includes the  $p$  itself, the other proper itineraries and a virtual itinerary that models passengers spilled.

(2) Decision variables

$q_p^r(x, \omega)$ : Number of passengers who are originally in itinerary  $p$  but ultimately served on itinerary  $r$  based on first-stage decision  $x$  under scenario  $\omega$ .

$k_i^e(x, \omega)$ : fleet assignment variable based on first-stage decision  $x$  under scenario  $\omega$ , equals 1 if type  $e$  is assigned to  $i$ , otherwise 0.

Let  $y(x, \omega)$  denote the above decision variables in the second-stage model for expressional simplicity.

The recourse model reflects the passenger arrival delay during the entire  $T_1$ . Notice that not only the flight cycles in the first-stage are considered, but also those re-constructed (combining re-timing and re-fleeting) flight cycles in the future. The objective function of the recourse model is defined

$$Q(x, \omega) := \min Q(x, \omega) = \min \sum_{p \in P} \sum_{r \in R(p, x, \omega)} d_p^r \times q_p^r(x, \omega) \quad (6)$$

$$\sum_{r \in R(p, x, \omega)} q_p^r(x, \omega) = n_p \quad \forall f \in F, \forall p \in P: \delta_f^p = 1 \quad (7)$$

$$\sum_{p \in P} \sum_{r \in R(p, x, \omega)} \delta_f^r \times q_p^r(x, \omega) \leq \sum_{e \in \epsilon} \sum_{i \in FC} SEAT_e \lambda_f^i k_i^e(x, \omega) \quad \forall f \in F \quad (8)$$

Constraints in Eqs. (7), (8) are the passenger re-accommodation constraints. They are referred to Bratu and Barnhart's model in Ref. [6]. Eq. (7) ensures that all passengers will arrive at their destinations finally. Eq. (8) requires that the number of passengers that transferred to a flight leg cannot exceed its seat capacity, and if the flight leg is cancelled, no passenger will be transferred to it. Notice that for  $\forall i \in FC_1$ ,  $k_i^e(x, \omega) = k_i^e$  because all the parameters are known in the first-stage model.

There is an implicated hard work on the recourse model, which is the generation of  $R(p, x, \omega)$ . The generation work is based on the eligible re-constructed flights. Compared with the re-scheduling solutions in the first-stage model, the flight re-construction for the second-stage is much more complicated under one scenario. All the eligible flights must meet the following requirements.

(1) The minimum turn-around time for the consecutive flights in one aircraft routing.

(2) No flight can fly before its STD.

(3) The arrival of flight cycles in each future time stage cannot exceed the corresponding arrival capacity in the hub.

(4) For the flight to fly, an aircraft should be assigned.

(5) At the end of  $T_1$ , there are enough aircraft available of each fleet type for the following regular flight schedule.

Theoretically, for each feasible first-stage solution and each scenario, a second-stage resource solution exists since cancellation strategy can always be applied to recover the flights, so the stochastic model has relatively complete resource.

### 3 Algorithm

The deterministic flight recovery model is already NP-hard, so obviously when it is extended to stochastic problem, more significant computational effort will be required. The proposed two-stage stochastic model has some special features as follow.

(1) The first-stage solution number is finite and can be enumerated. Regardless of the propagation effect on the recourse model, for given  $FC_1$ , airport capacity and the available aircraft, the combination of decision variables  $w_i$ ,  $u_i$ ,  $v_i$  and  $k_i^e$  in the first-stage are not hard to obtain.

(2) The objective function value of the recourse model  $Q(x, \omega)$  cannot be easily calculated. It has two layers; One is the schedule re-construction work on the flight schedule; the other is the passenger re-accommodation optimization based on the re-constructed flight schedule.

(3) The schedule re-construction solutions for the second-stage model are too large to enumerate even for one given disruption scenario. In our problem, the re-construction solutions in each stage depend on the solutions of the previous stages. As the stage goes further, the solutions number will increase exponentially. Besides, generating a feasible re-constructed flight schedule is time-consuming since the complicated constraints mentioned in the previous section. Thus, the resource model is not easy to solve.

(4) The passenger itinerary recovery work in the resource model can be solved quickly for given  $x$ ,  $\omega$ , and the re-constructed flight plan. (The detail algorithm to solve the model will be elaborated later)

For stochastic programs with large solution space of recourse model, a number of sampling based approaches have been proposed. Random sampling methods are used to obtain the statistical estimates of the expected value function. They can be classified into two groups: interior sampling and exterior sampling methods. Sample average approximation (SAA) method is one of the exterior methods where the sampling and optimization are decoupled. The basic idea is quite simple; random samples of scenario are generated and the expected value function is approximated by the corresponding sample average function<sup>[22]</sup>. More information on the SAA method can be found in Ref. [22]. Due to the requirement of short response time for recovery problem, an algorithmic framework is proposed to combine SAA

method and heuristic algorithm, which is referred to the Sölveling's approach in Ref. [16], to obtain good solutions in tractable computing time. The basic idea of the algorithm is listed as follows, and the details of the key techniques will be described in the following subsections.

(1) Enumerate the feasible decisions on flight cycles in the first-stage model.

(2) Index each first-stage solution as a node within the resulting decision tree.

(3) For each node, evaluate a sample set of candidate paths originating at the node via random sampling.

(4) Solve the corresponding SAA problem to obtain the upper bound for each node.

(5) Heuristically compute the lower bound for each node.

(6) Prune the nodes that are not promising.

### 3.1 Upper bound estimation

For the upper bound estimate, clearly the objective value of any feasible solution will be the upper bound for the optimal value  $Z^*$ . So suppose  $\hat{w}_i$  and  $\hat{k}_i^e$  denote one feasible solution in the first-stage model, i. e.,  $\hat{x} = \{\hat{w}_i, \hat{k}_i^e\}$ , and  $N$  the sample size, then the upper bound estimate can be defined as

$$U(Z^*) = \sum_{i \in FC_1} (\hat{w}_i c_i^{\text{cancel}} + \sum_{e \in \epsilon} \hat{k}_i^e \cdot c_i^e) + \sum_{n=1}^N Q(\hat{x}, \omega^n) / N \quad (9)$$

In the standard SAA method, it is natural that samples are generated from disruption scenario set  $\Omega$  according to probability distribution  $p$ .  $\Omega$  is finite but the re-constructed schedule solutions for a given  $\omega$  are too large to traverse, so  $Q(\hat{k}, \omega^n)$  is hard to determine. Therefore, the re-constructed schedule solutions can be considered as a re-construction set for given  $x$  and  $\omega$ , then sampling from it can obtain optimal or suboptimal  $Q(x, \omega)$  as Clarke did in large-scale deterministic recovery problem<sup>[23]</sup>. Since any feasible solution  $\hat{y}(\hat{x}, \omega^n)$  in second-stage model will compute the upper bound  $U(Q(\hat{x}, \omega^n))$  for  $Q(\hat{x}, \omega^n)$ , Eq. (10) can be defined to obtain the upper bound for  $Z^*$  in short time

$$\hat{z}_N = \sum_{i \in FC_1} (\hat{w}_i \cdot c_i^{\text{cancel}} + \sum_{e \in \epsilon} \hat{k}_i^e c_i^e) + \sum_{n=1}^N U(Q(\hat{x}, \omega^n)) / N \quad (10)$$

Eq. (11) is defined to update  $U(Q(\hat{x}, \omega^n))$ , where  $j$  represents the index of sampling from reconstruction set. The equation means when a better solution is obtained in the sampling process, the upper bound will be updated accordingly. As the sample size increases, the upper bound will decrease in piecewise way.

$$U_j(Q(\hat{x}, \omega^n)) = \min\{U_{j-1}(Q(\hat{x}, \omega^n)), Q_j(\hat{x}, \omega^n)\} \quad (11)$$

### 3.2 Passenger itinerary recovery algorithm

For given  $(\hat{x}, \omega^n)$  and one re-construction sample, the candidate recovery itineraries  $R(p, x, \omega)$  will be trivial to generate. The resource model (Eqs. (6)–(8)) will be optimized to obtain the passenger delay cost. For such problem, a heuristic method is used for the following three reasons: Firstly, the optimization problem is hard to solve and time-consuming to get the exact optimal result. Secondly, there is trade-off between computing time and solution quality. By saving time, more samples can be generated and it will improve the estimator quality in turn. Thirdly, the heuristic manner is more intuitive and acceptable for passengers in real world operations.

In each stage, the passenger itineraries are ordered by their "value", which was defined as how much they paid on their itineraries. High-value passengers will have higher priority in re-accommodation process. Since passengers are clustered in different itineraries in the paper, the disrupted passengers in highest priority itinerary will be re-accommodated first to the least delayed available itinerary. The algorithm can always obtain satisfactory solution because spilled passengers can transferred to the virtual itineraries. This greedy process will perform continuously until all passengers confirmed their new itineraries, and the delay cost is trivial to compute then. The detail of the passenger itineraries recovery algorithm is listed as follow.

```

Let  $Q(\hat{x}, \omega^n) = 0$ 
Sort the set  $F$  in according with the ascending STD
for  $f$  in the sorted set of  $F$ 
    search to find the involved itineraries and generate a
    value-descending itinerary set  $P_f$ 
    for  $p \in P_f$ 
        generate  $R(p, x, \omega)$  and sort it according to the
        ascending ETA
        while the number of passengers who has not been
        re-accommodated  $nb_p > 0$  do
            if  $R(p, x, \omega)$  cannot find available seats ex-
            cept for virtual cancellation ones
                update  $nb_p = 0$ 
                update  $Q(\hat{x}, \omega^n)$ 
            else
                greedily transfer the passengers of itiner-
                ary  $p$  to the  $R(p, x, \omega)$ 
                update  $Q(\hat{x}, \omega^n)$ ,  $nb_p$  and the seats infor-
                mation
            end while
        end for
    end for

```

### 3.3 Lower bound estimation

In our problem, the expected value function  $E[Q(x, \omega)]$  is approximated by the sample average function  $\sum_{n=1}^{N'} Q(x, \omega^n) / N'$ . Referrin to standard SAA method, the stochastic problem

$$Z_{N'} = \min \sum_{i \in FC_1} (w_i \cdot c_i^{\text{cancel}} + \sum_{e \in \epsilon} k_i^e \cdot c_i^e) + \sum_{n=1}^{N'} Q(x, \omega^n) / N' \quad (12)$$

is defined as the SAA problem and can be solved as a deterministic optimization algorithm. The SAA method proceeds by solving the SAA problem as Eq. (12) repeatedly. By generating  $M$  independent samples, each of them with size  $N'$ , and solving the associated SAA problems, one can obtain objective values  $Z_{N'}^1, Z_{N'}^2, \dots, Z_{N'}^M$ . Let

$$L(Z^*) = \sum_{m=1}^M Z_{N'}^m / M \quad (13)$$

where  $L(Z^*)$  denotes the average optimal objective function value for the  $M^*$  SAA problems. Since  $E[L(Z^*)] \leq Z^*$  [24],  $L(Z^*)$  provides a statistical estimate for a lower bound on the optimal value of the true problem.

Although Eq. (12) has already reduced the problem scale, it is still not easy to solve and will be time-consuming. As mentioned before, the



disruption scenario  $\Omega$  and the first-stage solutions set  $X$  are finite, so the combination (Cartesian product) set of  $\Omega \times X$  is also finite. For given  $x$ , the objective function on the first-stage model is easy to obtain since it is linear. Meanwhile, the minimization on sample average function  $\sum_{n=1}^N Q(x, \omega^n) / N'$  can be degenerated as  $N'$  minimization problem  $Q(x, \omega)$ . Therefore, the heuristic equation in Ref. [23] can be applied to intuitively express the lower bound for  $Q(x, \omega^n)$ , which is

$$L_{j+1}(Q(x, \omega^n)) = L_j(Q(x, \omega^n)) + [U_{j+1}(Q(x, \omega^n)) - L_j(Q(x, \omega^n))] \cdot \{[Pr_{j+1}(Q(x, \omega^n)) + 1] / 2\}^K \quad (14)$$

where  $j$  represents the index of sampling from reconstruction solutions. When  $j \rightarrow \infty$ ,

$\lim_{j \rightarrow \infty} Pr_{j+1}(Q(x, \omega^n)) = 1$ , so

$\lim_{j \rightarrow \infty} \{[Pr_{j+1}(Q(x, \omega^n)) + 1] / 2\}^K = 1$ , thus  $L_{j+1}(Q(x, \omega^n))$  will asymptotically equal  $U_{j+1}(Q(x, \omega^n)) \cdot \{[Pr_{j+1}(Q(x, \omega^n)) + 1] / 2\}^K$  is used instead of simple form  $Pr_{j+1}(Q(x, \omega^n))$  because when  $j$  is small, the lower bound estimator can increase more quickly. The parameter  $K$  controls the increasing step. It is obvious that  $L_{t_0}(x)$  can set to be 0.

To avoid identical samples generated, a non-replacement way to get re-construction samples is chosen. Suppose the number of feasible solutions in the recourse model for given  $(x, \omega^n)$  is  $S(x, \omega^n)$ , and only one of them is optimal solution, then  $Pr_j(Q(x, \omega^n))$  denotes the probability that in  $j$  samples, the optimal solution will be obtained. By simple knowledge of probability,  $Pr_j(Q(x, \omega^n)) = j / S(x, \omega^n)$ . For large-scale problem, where  $S(x, \omega^n)$  cannot be obtained accurately, the value can be estimated based on the reconstruction solution tree structure.

### 3.4 Termination criteria

The most common termination criterion for SAA method is the optimal gap, which equals  $U(Z^*) - L(Z^*)$  in our problem. Since all the first-stage solutions can be enumerated and their costs can be computed in advance, more work will focus on the second-stage model. The

bounds will be used to evaluate the first-stage solution. Thus, it is obvious that the algorithm will stop when the upper bound of certain solution is lower than the lower bounds of the other solutions, then the most promising one will be obtained.

## 4 Computational Study

A small case was generated based on the operational data from a Chinese airline. The flight network held 6 airports ( $N_{\text{apt}} = 6$ ), which consisted of 5 spoke airports and one hub airport. There were 47 flight legs, which compose 23 flight cycles, operated by 9 aircraft of 2 types (with MTT 40 and 60 min, and seat capacity of 200 and 250, respectively) in this case. 7 797 passengers who were grouped in 82 itineraries traveled in the flight schedule network from 08:00 to 02:00 in the next day.  $T_0$  was defined as 09:00–13:00,  $T_1$  08:00–18:00, and  $T_2$  08:00–02:00 in the next day. If some passengers could not be re-accommodated during  $T_2$ , they would be re-directed to the virtual spilled itinerary. The other relative parameters and their values are listed in Table 1. The algorithmic framework was implemented in C++ and Python on a laptop with 4 GB installed RAM and i5-3317U CPU 1.70 GHz.

For the first-stage, the capacity of the hub was known. Three disruption scenarios were simulated to represent minor, medium, severe weather conditions which would reduce the hub capacity by 25%, 50% and 75%, respectively. To compare different weather conditions, different probabilities on the three conditions were set to generate different sets  $\Omega_A$  and  $\Omega_B$  as shown in Table 2.

Table 1 Parameters in the small test instance

Parameter	Value
Minimum connection time for passengers/min	30
The length of unit time period/min	60
Cost per passenger per minute/min	\$ 1
$K$	10
$N$	1 000

**Table 2** Probability of scenario set

Scenario set	$\omega_{\text{minor}}$	$\omega_{\text{medium}}$	$\omega_{\text{severe}}$
$\Omega_A$	0.5	0.3	0.2
$\Omega_B$	0.2	0.3	0.5

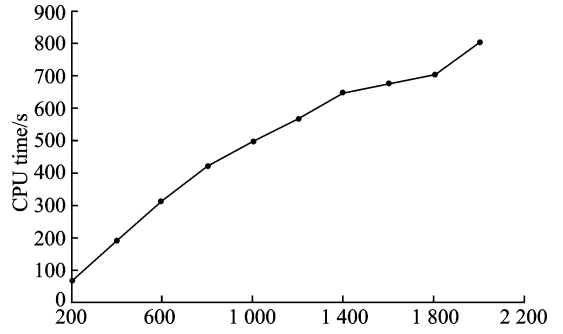
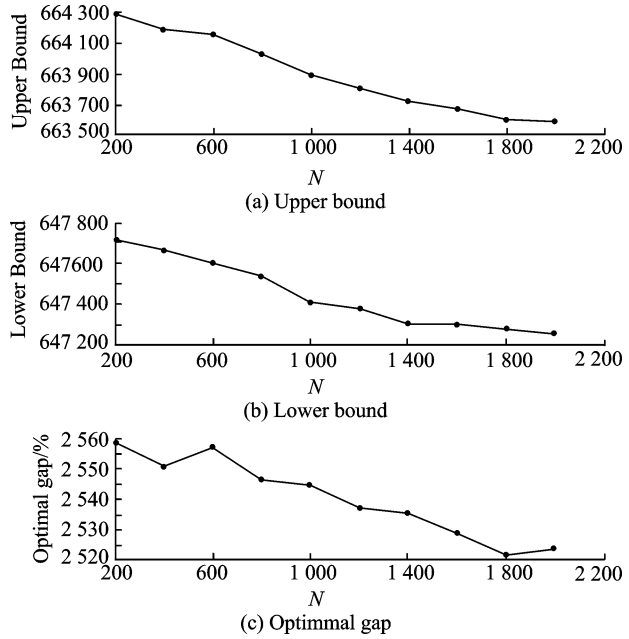
The first-stage model has 3 solutions (represented by  $x_1, x_2, x_3$ ), and the second-stage model will be used to evaluate the influence of those solutions under stochastic disruptions. Table 3 lists the computational results of the two different scenario sets. LB, UB, and OG represent lower bound, upper bound, and optimal gap, respectively. Since  $\Omega_A$  had a more positive weather condition forecast than  $\Omega_B$ , the objective values of their three solutions exhibited much lesser cost than the counterparts of the latter ones. The optimal gap for each solution was around 5%, which was good enough for operational problems. The two problems both chose first-stage solution  $x_1$ , which indicated that under such two different uncertainty scenario sets, the performances of  $x_1$  were both promising. It was reasonable because  $x_1$  represented the decision with least deviation of the original schedule.

**Table 3** Computational results for the small case

Scenario set	Numerical item	$x_1$	$x_2$	$x_3$
$\Omega_A$	LB	647 411.70	660 165.74	660 512.50
	UB	663 885.08	684 749.87	686 551.63
	OG/%	2.54	5.66	5.93
$\Omega_B$	LB	893 274.23	911 118.45	911 428.38
	UB	930 998.53	938 849.02	940 433.53
	OG/%	4.22	5.10	5.28

To study the computing time of the algorithm, different values of sample size  $N$  were set to test the case under  $\Omega_A$ . Fig. 3 shows the average CPU times for the problem with sample size  $N \in \{200, 400, \dots, 2000\}$ . It indicates that the computing time is almost linear to the sample size. As long as the computing time is shorter than the unit time period, the recovery process can be conducted dynamically to accommodate airlines daily operation under stochastic circumstances.

Fig. 4 shows the upperbounds, lower bounds,

Fig. 3 Average CPU times for different  $N$ Fig. 4 Upper bound, lower bound, and optimal gap for different  $N$ 

and optimal gaps for different sample sizes range from 200 to 2000. Figs. 4 (a, b) show that the upper bounds and lower bounds will decrease monotonously when  $N$  increases, which indicates that better solutions will be obtained with larger  $N$ . Fig. 4 (c) shows the optimal gap also has the decreasing trend although there are some outliers due to the stochasticity. Considering the CPU time performance showed in Fig. 3, it is obvious that there is trade-off between computing time and the solution quality. Decision makers can choose appropriate sample size based on the available response time when irregularity happens.

## 5 Conclusions

In this paper, flight recovery problem combi-

ning flight timetable, fleet re-assignment and passenger re-accommodation under uncertain hub capacity was studied. It was modeled as a flight cycle oriented two-stage integer stochastic problem. Since the model had some special structure features, an algorithmic framework based on sample average approximation and greedy heuristic methods was designed. The case from a Chinese airline was chosen to test the method. The results showed that for a small sample size, the optimal gap was around 5%, which satisfied operational problems. The computing times were also tracked and they were linear to the sample size. Thus, the proposed method showed its ability of obtaining satisfying solution in tractable time.

There are some interesting problems raised for future work during the research. Theoretically, if parallel computing technique is used, every first-stage model solution can be sent to different CPUs, and with high-speed computer, the computing time can be controlled to be shorter than a smaller time period, like 5 min. This requires large scales of tests to calibrate and evaluate. The disruption scenario set is assumed to be finite in this paper, and this situation can be extended into a scenario tree to represent more precise stochastic information on the disruption.

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