# Influence of Helix Angle on Stability of Milling Flexible Parts

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Abstract: Geometry of end mill cutters plays a vital role in the stability of the flexible parts milling. The present study considers helix angle and number of flutes. Multi-frequency solution is established to draw stability Lobe diagram(SLD) for different helix angles and number of flutes. SLD for the two cases shows that the greater the value of helix angle, the more stable the milling process will be, and conversely increasing the number of flutes degrades the stability of flexible parts milling. A simple empirical methodology is adopted to employ the inclined plane workpiece geometry offering a gradual increase of the axial depth of cut in the feed direction. Surface roughness is used as a measure of stability. Test results corroborate the model conclusions very well.

Key words: helix angle; flute; end mill cutter; flexible parts milling; stability

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## 0 Introduction

High efficiency products in modern industries like aerospace, aircraft and automobile guarantee the needed components with thin sections, which can greatly reduce structural stiffness. However, the machining of these thin sectioned parts presents a very difficult condition to keep stability. With HSM the metal removal rate increased dramatically through a combination of large axial depths of cut and high spindle speeds. One limitation on the allowable axial depth is regenerative chatter. Chatter is the most obscuring phenomenon and considerable research has been documented regarding the prediction, control and elimination of chatter.

Many out-of-process and in-process techniques have been established to suppress, control and predict chatter. SLD stands at the border between a stable cut (i. e. no chatter) and an unstable cut (i. e. with chatter) in terms of the axial depth-of cut as a function of the spindle speed. Research in the area of chatter dates back to as early as 1906 when Taylor stated that chatter is

the "most obscure and delicate of all problems facing the machinist"<sup>[1]</sup>. Other pioneering works in the field of chatter vibration are nominated to Tlusty et al. <sup>[2]</sup>, Merritt<sup>[3]</sup>, Sridhar et al. <sup>[4]</sup>. A study shows that presently Altintas, Budak and Tlusty are the key stakeholders in the business of chatter<sup>[5]</sup>.

Remarkable contribution in the research of workpiece vibrations in thin walled parts have been made by Budak<sup>[6-7]</sup>, Seguy et al. <sup>[8]</sup>, Thevenot<sup>[9]</sup>, Song et al. <sup>[10]</sup>, Tang and Liu<sup>[11]</sup>, Davies and Balachandran<sup>[12]</sup>, and Adetoro et al. <sup>[13]</sup>. The influence of mill helix angle on stability was first considered by Zatarain et al. <sup>[14]</sup>. Later Zatarain et al. <sup>[15]</sup> analyzed the concept of directional factors for chatter stability in milling including the effect of the helix angle. Shirase and Altintas<sup>[16]</sup>, Budak<sup>[17-18]</sup>, Sims et al. <sup>[19]</sup>, Turner et al. <sup>[20]</sup>, Yusoff et al. <sup>[21]</sup> demonstrated in their work that with variable pitch and variable helix milling tools the stability of the milling improved by disrupting the regenerative effect.

In the study addressed here the milling sta-

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bility of flexible workpiece has been investigated under the influence of helix angle and number of flutes of end mill cutter by considering the improved helix angle model for flexible parts presented by Masud<sup>[22]</sup> in which very small changes in the initial and final immersion positions of the cutter arc segment were contemplated, to reflect the actual cutting conditions. Multi-frequency solution of the proposed model was obtained. Stability lobe diagram was constructed for variable helix angles and variable number of flutes. An increase in helix angle showed improved stability as compared to low helix angle values. Stability lobe diagram for variable number of flutes showed that the stability of the flexible parts milling increased with fewer number of flutes whereas increasing the number of flutes degraded the stability of flexible parts. A simple empirical methodology employing the inclined plane workpiece geometry was used to validate the model results. The inclined plane workpiece geometry offered the increase of axial depth of cut along the length of cut enabling varying stability condition for one spindle speed. Surface roughness was measured at different points along the cut as a gauge of stability of the cutting process, hence eliminating the need of sophisticated analytical tools and equipment like, impact hammers, piezoelectric transducers, specific software, and sensors for chatter detection.

# 1 Chatter Stability Model

As shown in Fig. 1, the milling cutter with N teeths and the workpiece are considered to obtain flexibility in feed (x-direction), normal (y-direction) and spindle (z-direction) directions.  $\varphi_j$  is the immersion angle of the jth teeth measured from the y-direction.  $\Omega$  (r/s) is the spindle speed. Tangential ( $dF_{t,j}(\varphi,z)$ ), radial ( $dF_{r,j}(\varphi,z)$ ), and axial ( $dF_{a,j}(\varphi,z)$ ) forces acting on a differential flute element with height dz (see Fig. 2) yield as

$$\begin{cases} dF_{t,j}(\varphi,z) = [K_{tc} \cdot h_j(\varphi_j(z)) + K_{te}] dz \\ dF_{r,j}(\varphi,z) = [K_{rc} \cdot h_j(\varphi_j(z)) + K_{re}] dz \end{cases} (1)$$
$$dF_{a,j}(\varphi,z) = [K_{ac} \cdot h_j(\varphi_j(z)) + K_{ac}] dz$$

where  $\varphi_j(z)$  is the immersion angle at the axial distance z.  $\varphi_j(0) = \varphi + j \cdot \varphi_p$ ,  $\varphi_j(z) = \varphi + j \cdot \varphi_p - k_\beta z$ ,  $k_\beta = (2\tan\beta)/D$ .  $\beta$  is the helix angle and D the effective diameter of the cutting tool. The dynamic chip thickness i

$$h_{j}(\varphi,z) = g_{i}(\varphi) \left[ (f + \Delta x + \Delta u) \sin \varphi_{j}(z) + (\Delta y + \Delta v) \cos \varphi_{j}(z) \right]$$
 (2)

where f is the feed per tooth,  $\Delta x = x(t) - x(t - T)$ ,  $\Delta y = y(t) - y(t - T)$  and  $\Delta u = u(t) - u(t - T)$ ,  $\Delta v = v(t) - v(t - T)$  are the dynamic displacements of the cutter and workpiece at the previous and the present tooth periods, respectively.

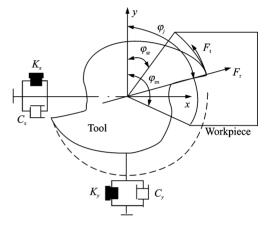


Fig. 1 Dynamic model of milling

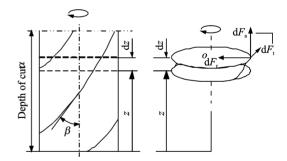


Fig. 2 Helical end mill and its thin disk element

The function  $g_i(\varphi)$  is a unit step function to determine whether the tooth is in or out of cut, that is

$$g_{i}(\varphi) = \begin{cases} 1 & \varphi_{st} \leqslant \varphi \leqslant \varphi_{ex} \\ 0 & \text{others} \end{cases}$$
 (3)

where  $\varphi_{\rm st}$  and  $\varphi_{\rm ex}$  are the start and the exit immersion angles of the cutter, respectively. The elemental forces in feed (x), normal (y) and axial (z) directions are

$$\begin{cases} \mathrm{d}F_{x,j}(\varphi_j(z)) = -\,\mathrm{d}F_{\mathrm{t},j} \bullet \cos\varphi_j(z) - \mathrm{d}F_{\mathrm{r},j} \bullet \sin\varphi_j(z) \\ \mathrm{d}F_{y,j}(\varphi_j(z)) = \mathrm{d}F_{\mathrm{t},j} \bullet \sin\varphi_j(z) - \mathrm{d}F_{\mathrm{r},j} \bullet \cos\varphi_j(z) \\ \mathrm{d}F_{\mathrm{a},j}(\varphi_j(z)) = \mathrm{d}F_{\mathrm{a},j} \end{cases}$$

Total cutting force produced by the flute is calculated by an integration of the differential cutting forces along the in-cut-portion of the flute j at each plane along z direction (5). Three distinct ways of interaction between immersion section of the cutter arc segment and a helical tooth j are shown in Fig. 3.

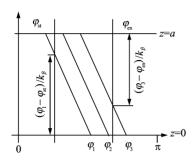


Fig. 3 Helical flute-part ace integration zones

$$F_{q,j} = \begin{cases} \int_{0}^{(\varphi_{i,j} - \varphi_{a})/k_{j}} \mathrm{d}F_{q,j} & \varphi_{1,j} \in [\varphi_{\mathsf{st}}, a \tan\beta + \varphi_{\mathsf{st}}) \\ \int_{0}^{a} \mathrm{d}F_{q,j} & \varphi_{2,j} \in [a \tan\beta + \varphi_{\mathsf{st}}, \varphi_{\mathsf{ex}}); q = x, y, z \\ \int_{(\varphi_{i,j} - \varphi_{a})/k_{j}}^{a} \mathrm{d}F_{q,j} & \varphi_{3,j} \in [\varphi_{\mathsf{ex}}, a \tan\beta + \varphi_{\mathsf{ex}}] \end{cases}$$

$$(5)$$

Assuming  $K_{\rm te}\!=\!K_{\rm re}\!=\!K_{\rm ae}\!=\!0$ , and neglecting the axial force, the following equations can be obtained

$$F_{x,j} = \begin{cases} \frac{g(\varphi_{1,j})K_t}{4k_\beta} \{(\Delta x + \Delta u + f) \left[K_r(\varepsilon_1 + \varepsilon_2) + \varepsilon_3\right] + \\ (\Delta y + \Delta v) \left[K_r\varepsilon_3 + \varepsilon_2 - \varepsilon_1\right] \} \\ \frac{g(\varphi_{2,j})K_t}{4k_\beta} \{(\Delta x + \Delta u + f) \left[K_r(\varepsilon_4 - 2k_\beta a) + \varepsilon_5\right] + \\ (\Delta y + \Delta v) \left[K_r\varepsilon_5 - \varepsilon_4 - 2k_\beta a\right] \} \\ \frac{g(\varphi_{3,j})K_t}{4k_\beta} \{(\Delta x + \Delta u + f) \left[K_r(\varepsilon_6 + \varepsilon_7) - \varepsilon_8\right] + \\ (\Delta y + \Delta v) \left[K_r\varepsilon_8 + \varepsilon_7 - \varepsilon_6\right] \} \end{cases}$$

$$(6a)$$

$$F_{y,j} = \begin{cases} \frac{g(\varphi_{1,j})K_{t}}{4k_{\beta}} \{(\Delta x + \Delta u + f) \left[K_{r} \varepsilon_{3} - \varepsilon_{2} - \varepsilon_{1}\right] + \\ (\Delta y + \Delta v) \left[K_{r} (\varepsilon_{2} - \varepsilon_{1}) - \varepsilon_{3}\right] \} \end{cases}$$

$$\frac{g(\varphi_{2,j})K_{t}}{4k_{\beta}} \{(\Delta x + \Delta u + f) \left[K_{r} \varepsilon_{5} + 2k_{\beta}a - \varepsilon_{4}\right] + \\ (\Delta y + \Delta v) \left[K_{r} (-\varepsilon_{4} - 2k_{\beta}a) - \varepsilon_{5}\right] \}$$

$$\frac{g(\varphi_{3,j})K_{t}}{4k_{\beta}} \{(\Delta x + \Delta u + f) \left[K_{r} \varepsilon_{8} - \varepsilon_{7} - \varepsilon_{6}\right] + \\ (\Delta y + \Delta v) \left[K_{r} (\varepsilon_{7} - \varepsilon_{6}) - \varepsilon_{8}\right] \}$$

where  $\epsilon_1 = \sin^2 \varphi_{1,j} - \sin^2 \varphi_{st}$ ,  $\epsilon_2 = 2\varphi_{st} - 2\varphi_{1,j}$ ,  $\epsilon_3 =$ 

$$\begin{split} &\cos^2\varphi_{1,j} - \cos^2\varphi_{\rm st} \; , \; \varepsilon_4 = \sin^2\varphi_{2,j} + \sin^2\left(k_{\beta}\alpha - \varphi_{2,j}\right) \; , \\ \varepsilon_5 = &\cos^2\varphi_{2,j} + \cos^2\left(k_{\beta}a - \varphi_{2,j}\right) \; , \; \varepsilon_6 = \sin^2\varphi_{\rm ex} \; + \\ &\sin^2\left(k_{\beta}a - \varphi_{3,j}\right) \; , \; \varepsilon_7 = 2\varphi_{3,j} - 2\varphi_{\rm ex} - 2k_{\beta}a \; , \; \varepsilon_8 = \\ &\cos^2\varphi_{\rm ex} - \cos^2\left(k_{\beta}a - \varphi_{3,j}\right) \; , \; K_{\rm t} = K_{\rm tc} \; , \; K_{\rm r} = \\ &K_{\rm rc}/K_{\rm tc} \; , \varphi_{i,j} = \varphi_i + j\varphi_p - K_{\beta}Z \; . \end{split}$$

The cutting forces contributed by all flutes are calculated and summed to obtain the total instantaneous forces on the cutter at immersion  $\varphi$ 

$$F_q = \sum_{j=1}^{N} F_{q,j} \quad q = x, y$$
 (7)

Converting Eq. (7) into matrix form as follow

$$\mathbf{F} = \frac{1}{4k_{\beta}} K_{\tau} B(t) \begin{pmatrix} \Delta x \\ \Delta y \end{pmatrix} + \frac{1}{4k_{\beta}} K_{\tau} \mathbf{C}(t) \begin{pmatrix} f \\ 0 \end{pmatrix}$$
(8)

where

$$\begin{aligned} \mathbf{F} &= \begin{bmatrix} F_x & F_y \end{bmatrix}^{\mathrm{T}} \\ B(t) &= \sum_{j=1}^{N} g(\varphi_{1,j}) B_1(\varphi_{1,j}) + \\ &\sum_{j=1}^{N} g(\varphi_{2,j}) B_2(\varphi_{2,j}) + \sum_{j=1}^{N} g(\varphi_{3,j}) B_3(\varphi_{3,j}) \\ \mathbf{C}(t) &= \begin{bmatrix} B(t)_{11} \\ 0 \end{bmatrix} \end{aligned}$$

Converting Eq. (6) from time domain to frequency domain

$$F(\omega) = \frac{K_{\text{tc}}}{4k_{\beta}} \left[ B^* (\omega) * \Delta(\omega) + f * C^* (\omega) \right]$$
(9)

where "\*" denotes the convolution integral.  $\Delta(\omega) = (1-\mathrm{e}^{-\mathrm{i}\omega T})\, \varPhi(\omega)\, F(\omega) \text{ is the displacement/regenerative vector; } \varPhi(\omega) \text{ is the sum of the "Frequency Response Function" (FRF) matrices of the tool/cutter and of the workpiece. The periodic directional matrix can be expanded into Fourier series$ 

$$B^{*}(\omega) = \sum_{r=-\infty}^{\infty} B_{r}^{*} \delta(\omega - n\omega_{T}) = R[B(t)] = \sum_{r=-\infty}^{\infty} B_{r}^{*} e^{in\omega_{r}t}$$
(10a)

$$\mathbf{B}^{*}(\omega) = \frac{N}{2\pi} \left[ \cdots + \begin{bmatrix} b_{xx}^{-1} & b_{xy}^{-1} \\ b_{yx}^{-1} & b_{yy}^{-1} \end{bmatrix} \delta(\omega + \omega_{T}) + \begin{bmatrix} b_{xx}^{0} & b_{xy}^{0} \\ b_{yx}^{0} & b_{yy}^{0} \end{bmatrix} \delta(\omega) + \begin{bmatrix} b_{xx}^{+1} & b_{xy}^{+1} \\ b_{yx}^{+1} & b_{yy}^{+1} \end{bmatrix} \delta(\omega - \omega_{T}) + \cdots \right]$$
(101)

Further multi-frequency solution is established to obtain the limiting depth of cut a as

$$a = \frac{2k_{\beta}}{K_{+}} \left[ \frac{\Lambda_{R,q}(1 - \cos(\omega_{c}T)) + \Lambda_{L,q}\sin(\omega_{c}T)}{1 - \cos(\omega_{c}T)} + \frac{\Lambda_{L,q}(1 - \cos(\omega_{c}T)) - \Lambda_{R,q}\sin(\omega_{c}T)}{1 - \cos(\omega_{c}T)} \right]$$

(11)

### 2 Simulations

For simulations purpose end mill cutter of 2 flute, 12 mm diameter,  $30^{\circ}$  helix angle end mill cutter was used. Dynamic parameters of cutter in X and Y directions are shown in Table 1. The

workpiece transfer function was identified in the normal direction (Y), in Table 2, as the dynamics in the feed direction (X) can be neglected due to its relative magnitude. The cutting coefficients of Al 7075,  $K_t = 999.28$  MPa,  $K_r = 426.4$  MPa were obtained<sup>[23]</sup>.

Table 1 Tool dynamic parameters in X and Y direction	Table 1	Tool dynamic	parameters in X	and Y	direction
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Mode/	Natural frequency $\omega_n/\mathrm{Hz}$		Modal stiffness $k / (N \cdot m^{-1})$		Modal damping ratio $\zeta/(10^{-2}\%)$	
Direction	X	Y	X	Y	X	Y
1	945.701 4	113 0.478 7	1.879 $7 \times 10^9$	5. $14 \times 10^7$	0.41	4.188
2	1 127.947 5	1 804.498	$1.709~0 \times 10^{8}$	7.488 $\times 10^7$	1.191	7.95
3	1 817.896 4	2 868.277 5	$5.95 \times 10^7$	$2.6308 \times 10^{7}$	11.4	6.413
4	2 845.67	3 701.345 6	$3.63 \times 10^7$	$2.0197 \times 10^7$	6.668	4.65
5	3 834.516 1	4 579.586	$2.2089 \times 10^{7}$	1.905 $\times$ 10 <sup>8</sup>	5.511 1	2.832

Table 2 Workpiece dynamic parameters in Y direction

_				
	Mode	Natural frequency $\omega_n/ ext{Hz}$	Modal stiffness k/ (N • m <sup>-1</sup> )	Modal damping ratio $\zeta / (10^{-2} \%)$
	1	3 173.992 5	1.192 $6 \times 10^7$	6.537 5
	2	3 248.687 4	1.160 $3 \times 10^9$	4.1127
	3	3 453.169 0	7.665 $7 \times 10^6$	3.446 6
	4	3 848.849 4	9.628 $0 \times 10^8$	4.961 5
	5	4 356.775 6	$1.662\ 1\times10^{7}$	5.3526

### (1) Helix Angle

Stability Lobe diagram (SLD) drawn for different helix angle is shown in Fig. 4. The simulation of the proposed model shows that helix angle affects the stability of flexible parts milling. An increase in helix angle shows improved stability and vice versa.

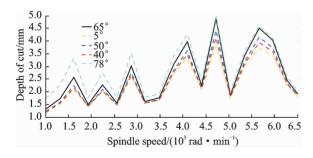


Fig. 4 SLD of helix angle

#### (2) Number of flutes

SLD drawn for different number of flutes is shown in Fig. 5. The simulation shows that increasing the number of flutes degrades the stability of flexible parts milling where as an end mill cutter with fewer flutes improves the stability of

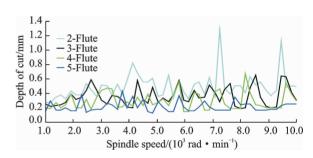


Fig. 5 SLD of number of flutes

the flexible parts milling.

# 3 Experimental Verification

In order to verify the simulation, a simple experimental methodology was adopted. A thin walled Aluminum 7075 workpiece 4 mm thick (Fig. 6) of inclined geometry offering a gradual increase of the axial depth of cut in the feed direction was used. Experimental setup is shown as Fig. 7. For each case of variable helix angle and number of flutes, two different values of end mill cutter were used. Test for each end mill cutter was performed with three sets of spindle speed, i. e. 2 000, 3 000 and 4 000 r/min on a three axis CNC milling machine. In order to ensure the same stability conditions among each test, following conditions were met:

- · Workpiece of same geometry was used.
- Torque wrench was used to apply the same clamping force on each test piece by clamping vice.
  - · Same tool holder was used and same over-

hang of the milling cutter was ensured.

• To analyze the two different cutters same cutting parameters i. e. spindle speed, feed per revolution and RDOC were ensured.

Surface roughness of the test specimen was measured as a Gauge to compare the stability of the two different cases. Stylus type surface roughness measuring instrument was used to measure the surface roughness of the test piece at different points along the cut. Large cut-off values i. e. 2.5 mm was employed to identify the chatter marks on the machined surface<sup>[24]</sup>.



Fig. 6 Inclined workpiece



Fig. 7 Experimental setup

#### (1) Helix angle

To verify the effect of helix angle on stability of thin walled workpiece, test were performed with two end mill cutters of dia 12 mm with helix angles 30° and 45° (Fig. 8). Tests were performed as listed in Table 3 along with surface roughness measurement made along the length of cut at two points 140 mm and 180 mm. Measurement was taken three times at a point and their mean was obtained. For each test RDOC was 1 mm and feed of 0.15 mm/tooth was used to maintain the constant chip load.

Table 3 Test parameters I

		Spindle	Feed	Surface rou	ghness/ $\mu$ m
Type 7	Γest	${\rm speed}/$	rate/	Point 1	Point 2
		(r•min <sup>-1</sup> )	) (mm•min <sup>-1</sup> )	(140 mm)	(180 mm)
30° helix	1	2 000	1 200	2.31	2.95
angle end	2	3 000	1 800	2.84	6.16
mill-cutter	3	4 000	2 400	4.61	5.33
45°helix	1	2 000	1 200	1.31	1.84
angle end	2	3 000	1 800	1.52	1.78
mill-cutter	3	4 000	2 400	1.66	3.75



Fig. 8 End mill cutter (helix angle)

#### (2) Number of flutes

In order to verify the effect of number of flutes of end mill cutter on stability of thin walled workpiece, test were performed with two 3-fluted and 4-fluted end mill cutters of dia 12 mm with helix angle 30° (Fig. 9). Tests were performed as listed in Table 4 along with surface roughness measurement made along the length of cut at two points 140 mm and 180 mm. Measurement was taken three times at a point and their mean was obtained. For each test RDOC was 1 mm and feed of 0.15 mm/tooth was used to maintain the constant chip load.



Fig. 9 End mill cutter (number of flutes)

Table 4	Test	parameters	I

		Spindle	Spindle Feed		Surface roughness/ $\mu$ m	
Type	Test	speed	rate/	Point 1	Point 2	
		(r•min <sup>-1</sup> )	(mm•min <sup>-1</sup> )	(140 mm)	(180 mm)	
3-fluted	1	2 000	900	1.8	2.2	
end mill-	2	3 000	1 350	2.1	3.5	
cutter	3	4 000	1 800	2.9	4.3	
4-fluted	1	2 000	1 200	2.31	2.95	
end mill-	2	3 000	1 800	2.84	6.16	
cutter	3	4 000	2 400	4.61	5.33	

### 4 Conclusions

Chatter in machining is the most debilitating problem faced by modern machining industries, and its prediction, elimination and control becomes even more critical when the part to be machined is flexible. The present research presents an important contribution in the stability analysis of flexible parts milling by studying the influence of helix angle and number of flutes of milling cutter after contemplating an improved helix angle model in which very small changes in the initial and final immersion positions of the cutter arc segment are considered. The study concludes that

- (1) A greater value of helix angle of the milling cutter improves the stability of the flexible parts milling and vice versa.
- (2) The smaller the number of flutes, the more stable the milling of flexible parts, will be and vice versa.

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